

Observatoire de Versailles Saint-Quentin-en-Yvelines





La mission Inspire-Sat Un mois en orbite

Un deuxième satellite d'observation de la Terre



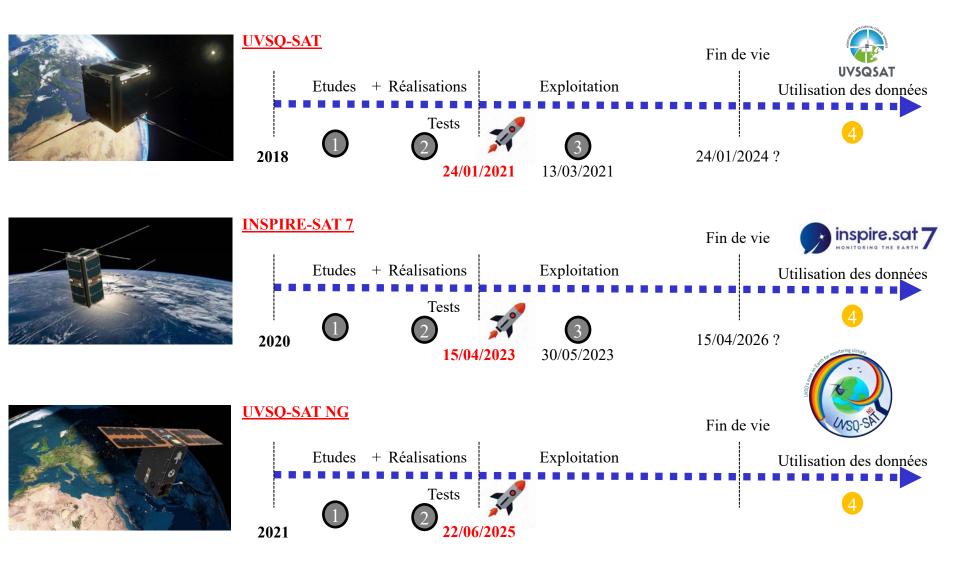
<u>13 mai 2023</u>



Dr. Mustapha MEFTAH [F4IXO]



1 – Timeline and objectives

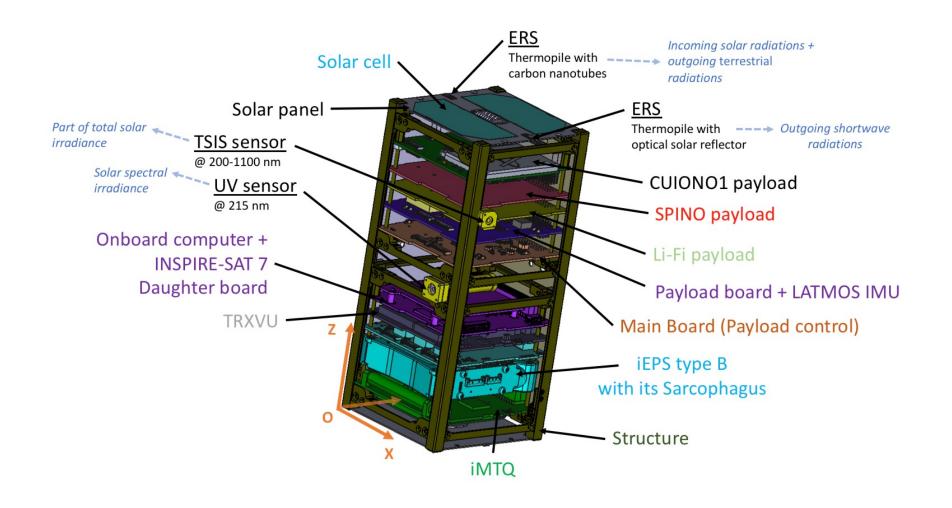


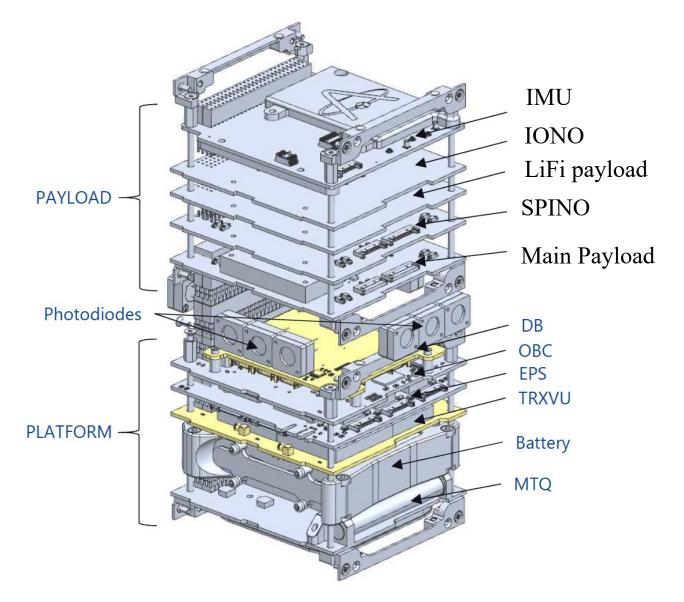
1 – Timeline and objectives

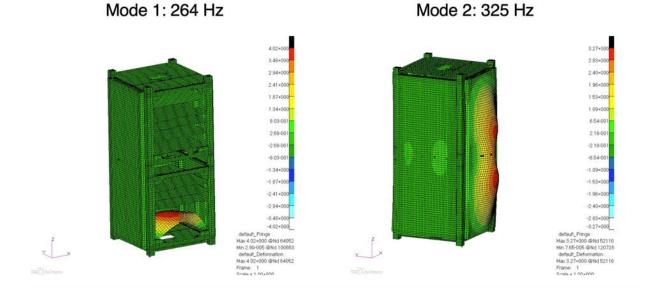
<u>Three main objectives in:</u>

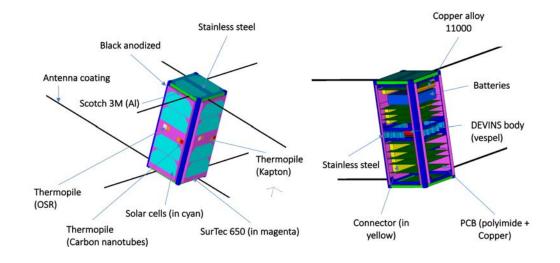
- <u>Science</u> : Earth observation (EO), Solar physics
- <u>Technology demonstration</u>: Instruments miniaturization for solar physics, Instruments validation & satellites constellation validation for Earth observations, UV sensors technology, Validation of an inertial measurements unit, Validation of the Totem electronic board, Validation of the SPINO radio-amateur payload, Validation of a Li-FI system, ...
- <u>Education & outreach</u> : Satellite development, Payload development, Software development, Training material

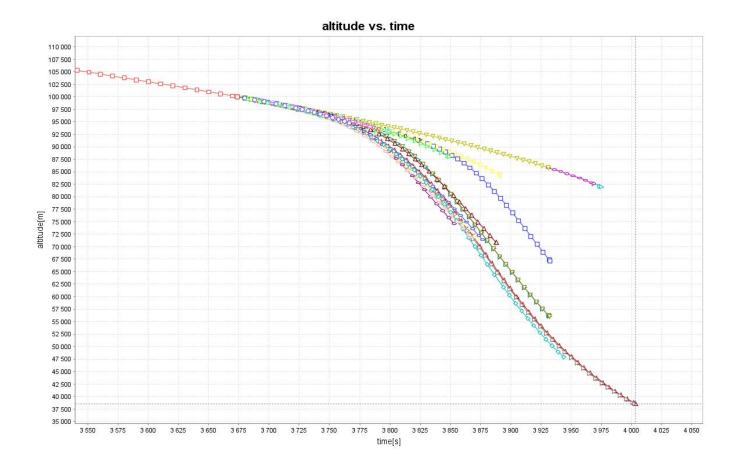


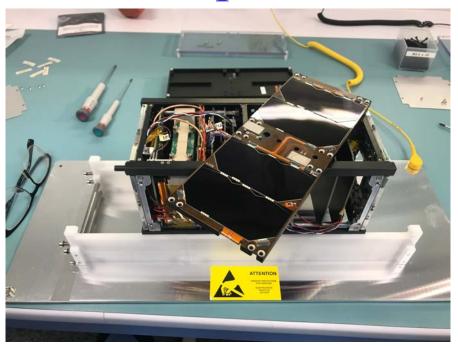


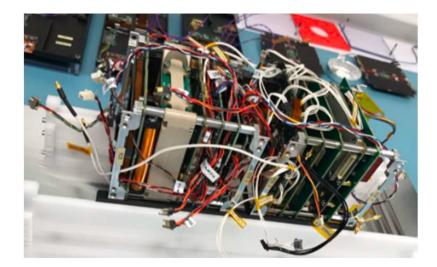


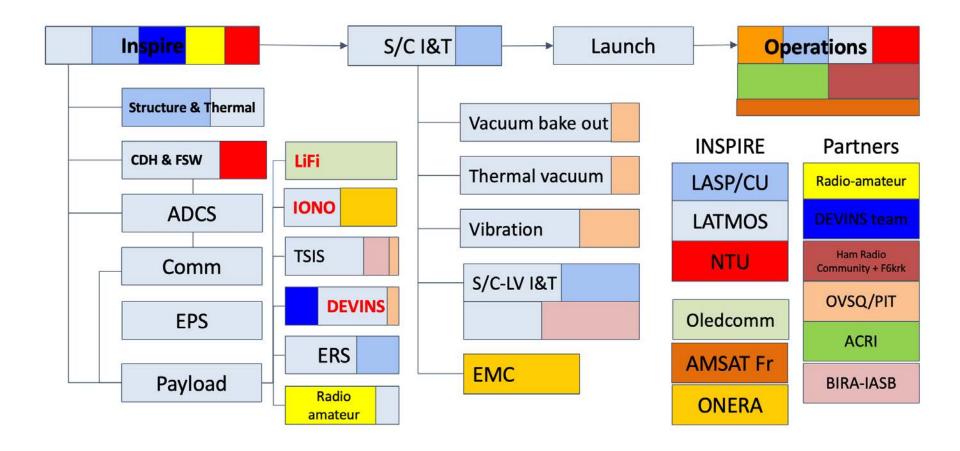






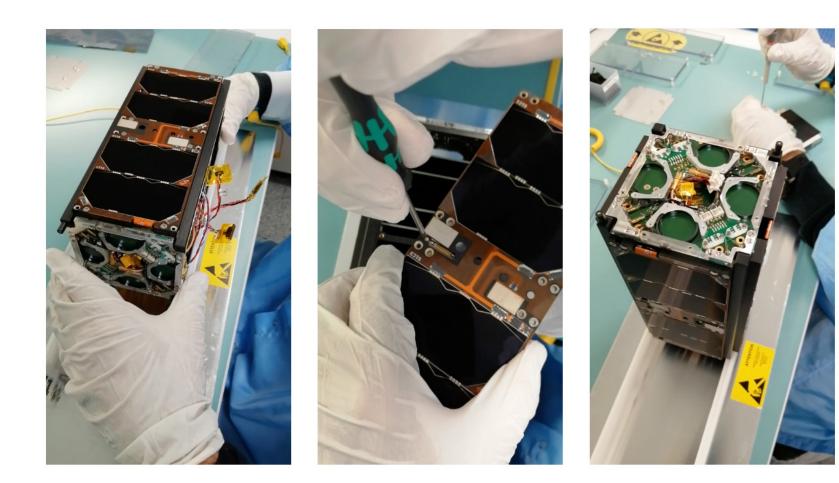


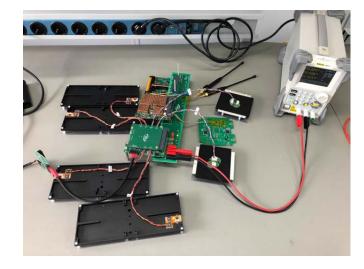




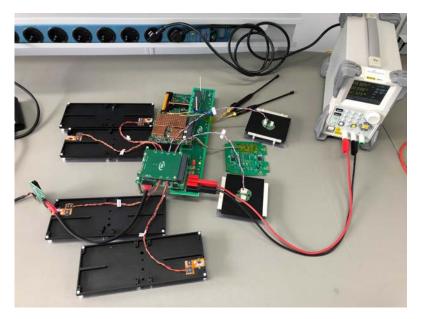
Main LATMOS/OVSQ facilities for INSPIRE-SAT 7

- 270 m² of ISO5, ISO7 and ISO8 clean rooms, equipped with permanent temperature and hygrometry monitoring, daily particles counting measurements. Several tools to facilitate the integration phases are available: space electrical components storage and packing, 3D and 2D mechanical metrology, clean soldering and gluing stations, trinocular visual inspection etc.
- A 40 kN shaker installed under an ISO5 laminar flux. This equipment is used for all prototypes mechanical validation as well as qualification and acceptance tests. Classical quasi-static, sine and random vibrations are achieved using this equipment.
- A 3 m³ thermal and optical vacuum chamber, installed in an ISO8 clean room. This chamber is able to simulate the Martian environment (regulated pressure steps between 5 and 20 mbar under CO2 atmosphere), as well as nominal space thermal vacuum environment below 10-5 mbar. Instruments up to 600x600x1600mm (50kg, 100W) fit into this chamber, also equipped with a cryogenic decontamination finger, a 300 amu mass spectrometer and a high spectral resolution McPherson type 225 monochromator allowing optical calibration within the 30-1200 nm range.



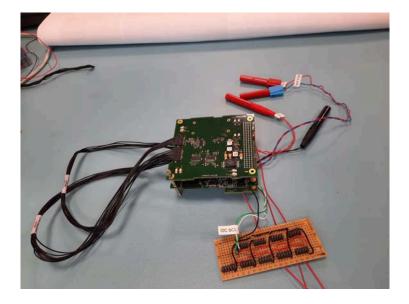




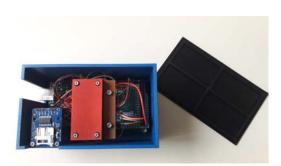




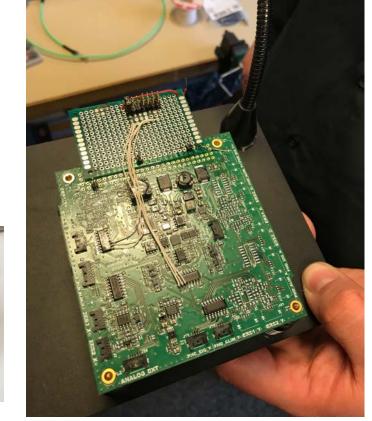




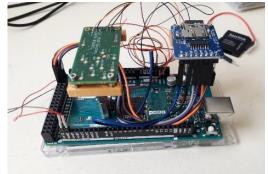


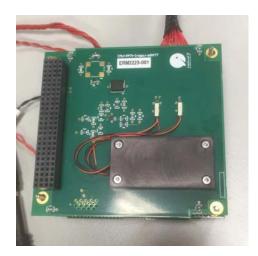




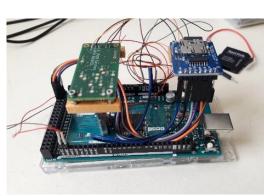


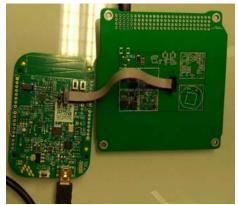


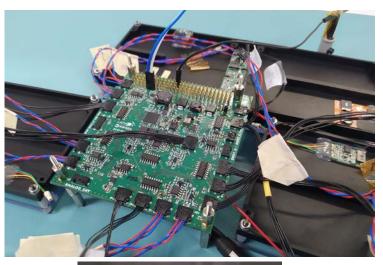






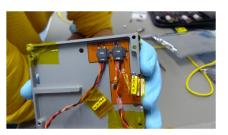












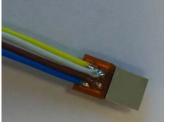


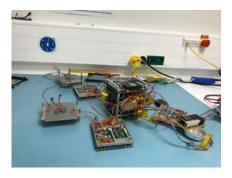


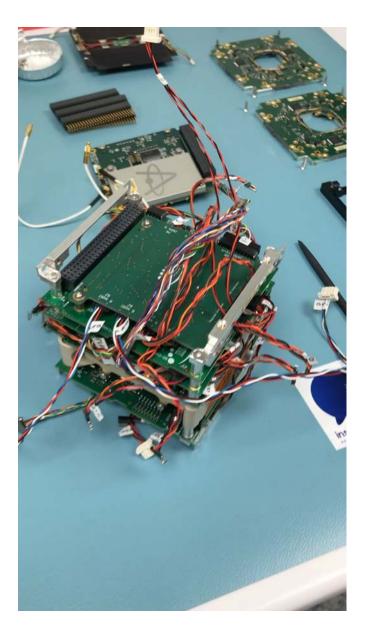




SITE PARIS-SACIA

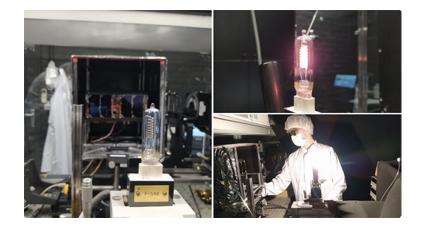


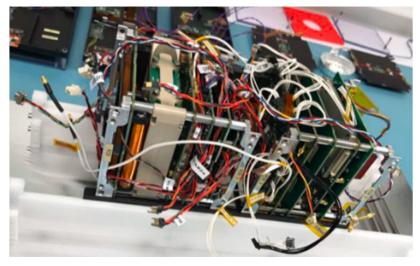




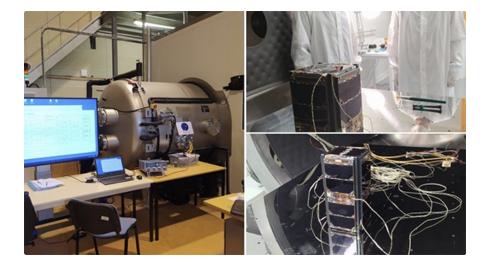


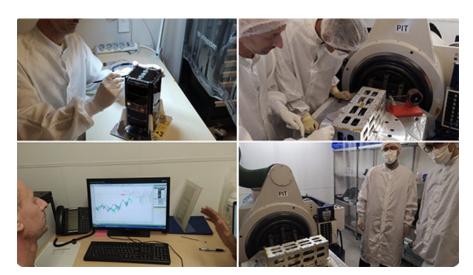




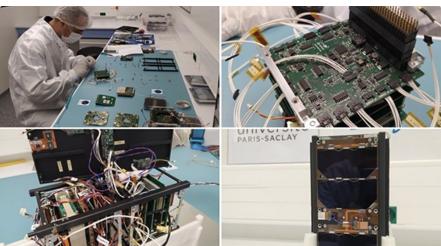


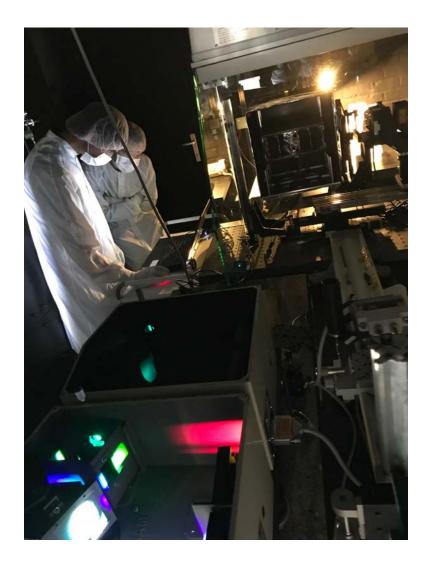


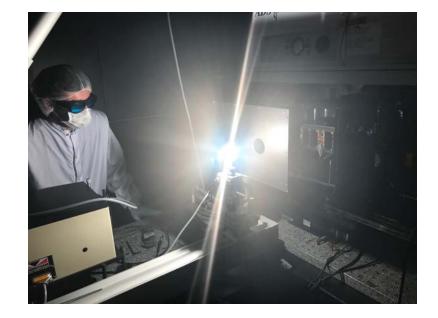


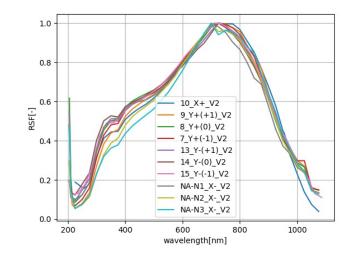


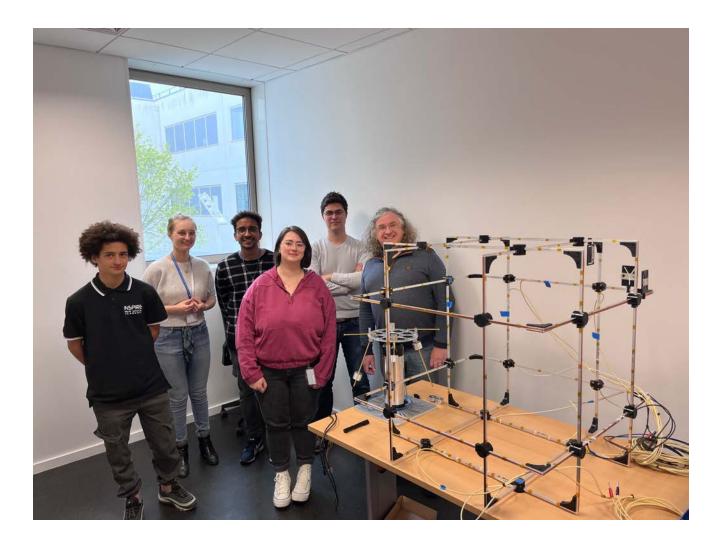


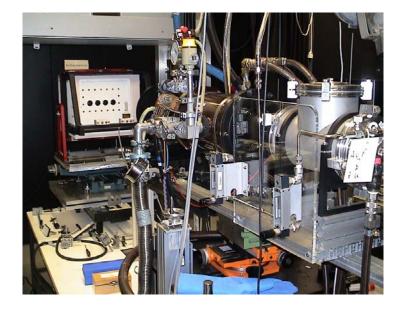








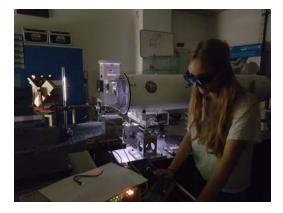


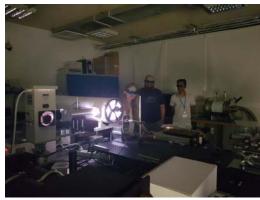


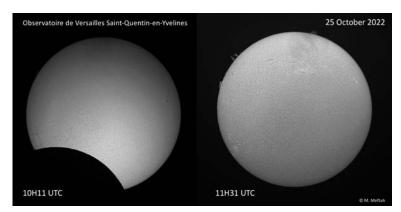






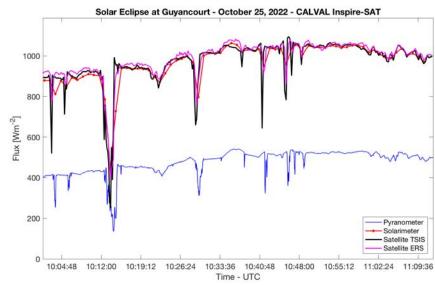
















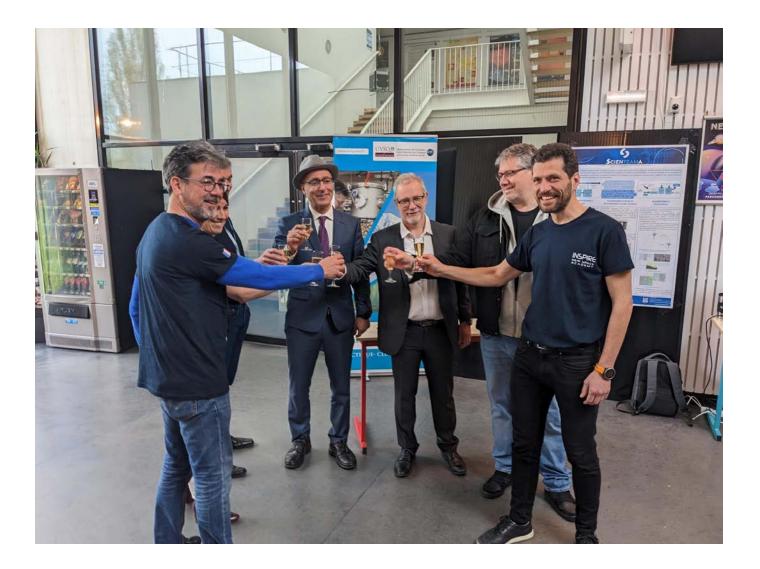
□ Launch with Transporter 7 – 15 April 2023

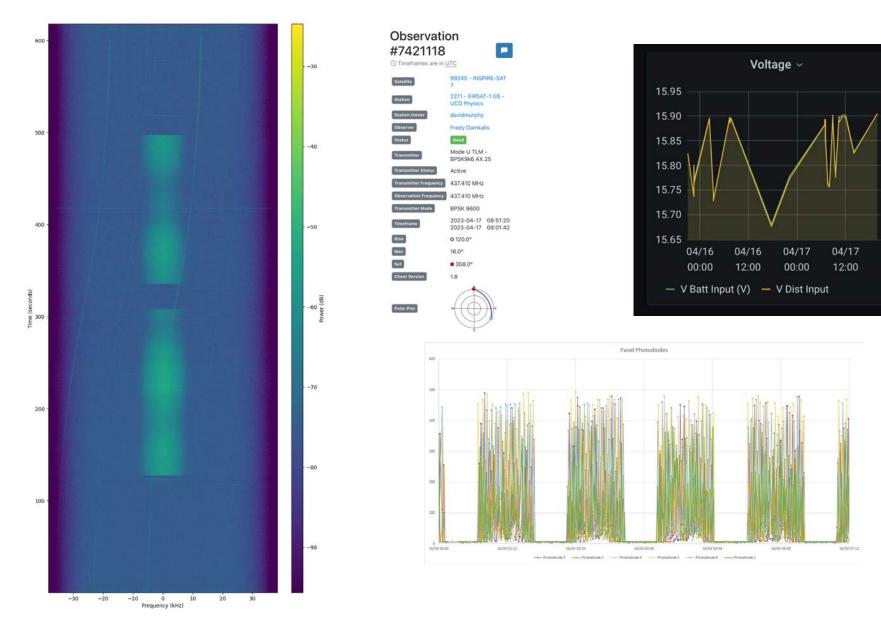


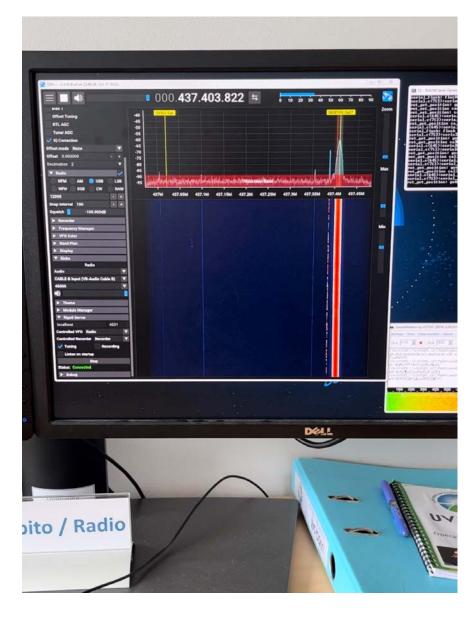




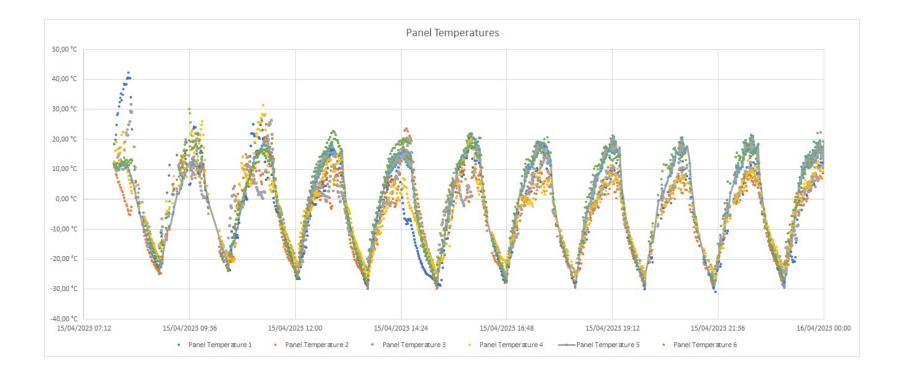


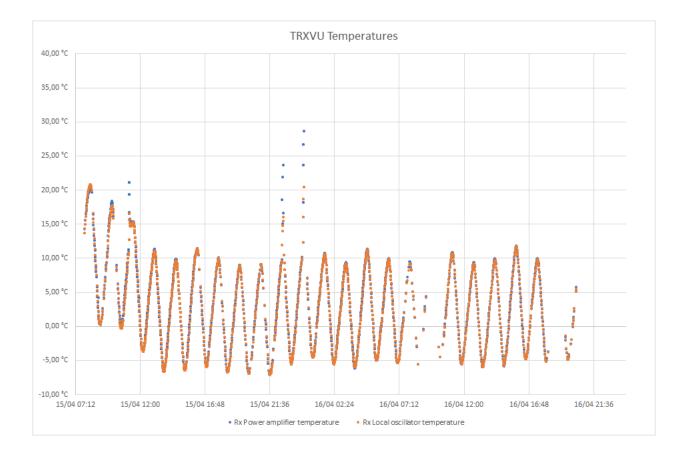


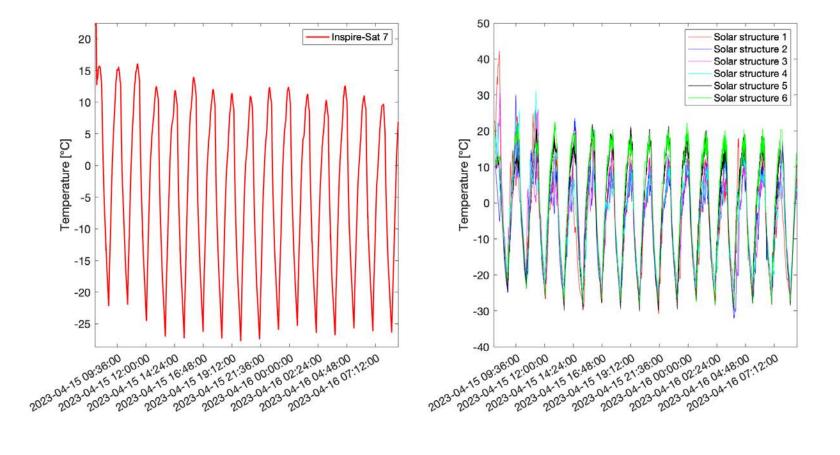


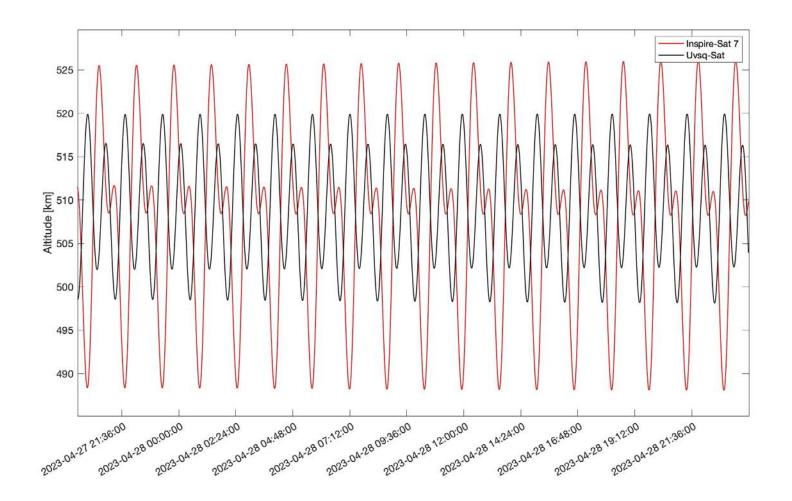












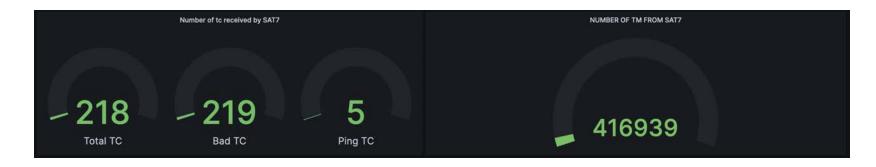
3 – Launch & commissioning phase

5 to 12 May 2023 – Inspire-Sat 7



3 – Launch & commissioning phase





	SOURCES SAT7			
			Value	Percent
		- LATMOS-to-Amsat	13747	33%
		- LATMOS-HERMES	12110	29%
		- ACRI-ST	6622	16%
		- EU1XX	5796	14%
		— Anonymous	2101	5%
		🗯 F4KMJ CS UPEC	692	2%
		🗕 F4KLK	13	0%

Mode U TLM - BPSK9k6 AX.25 Total Observations: 882 Mode V/U - FM Transceiver - SPINO		INSPIRE-SAT 7 NORAD ID 56211 Success Rate 48.41% Observations 886 429 101 345 11
Mode V/U - FM Transceiver - SPINO Total Observations: 4	Mode U TLM - BPSK9k6 AX.25	Total Observations: 882
	Mode V/U - FM Transceiver - SPINO	Total Observations: 4

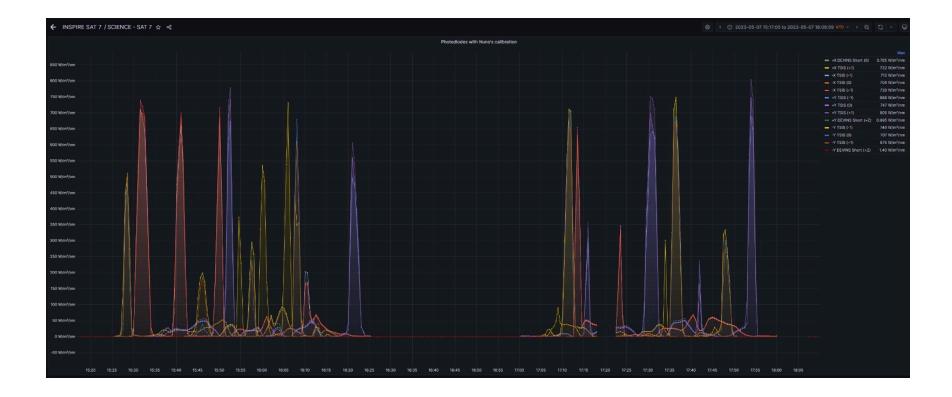
□ <u>ERS</u>



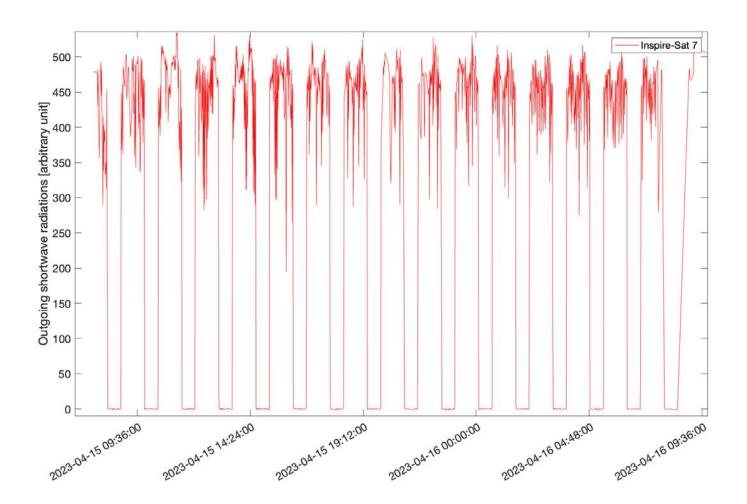
□ <u>ERS</u>



□ <u>TSIS & DEVINS</u>

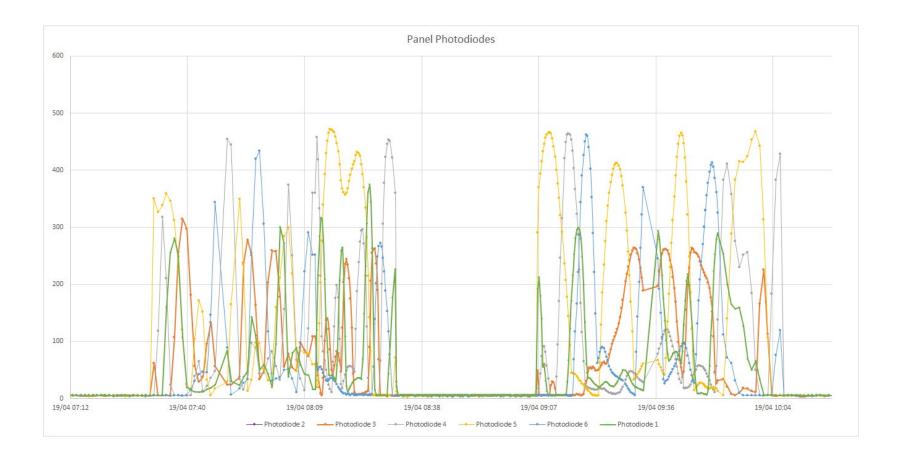


□ <u>OSR</u>



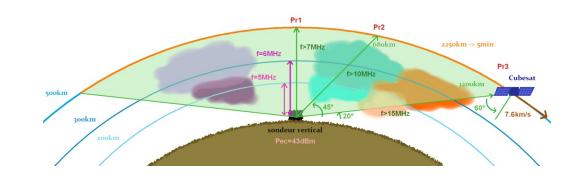
3 – Launch & commissioning phase

IMU – Before & after detumbling (Bdot algortihm)

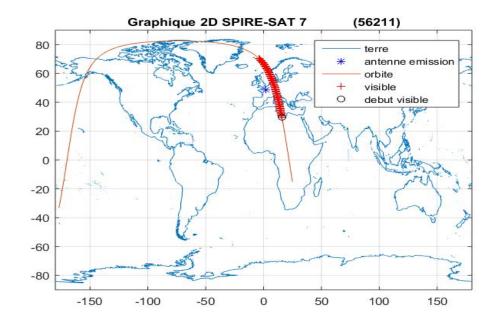


🗆 <u>LiFi</u>

Date	size of transferred data bits	speed in bps.	number of bit error	frames sent	frames received	latency in tens of ns	jitter in ns	mean digital voltage in mV	mean analog voltage in mV (N/A)	mean polarization voltage in mV (N/A)	mean current in mA (N/A)	peak current during transmission test in mA (N/A)	MCU temperature in °C	OLEDCOMM IC register 1 content	OLEDCOMM IC register 2 content	OLEDCOMM IC register 3 content	OLEDCOMM IC register 4 content	OLEDCOMM IC register 5 content	OLEDCOMM IC register 6 content	OLEDCOMM IC register 7 content	OLEDCOMM IC register 8 content
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□ <u>IONO</u>

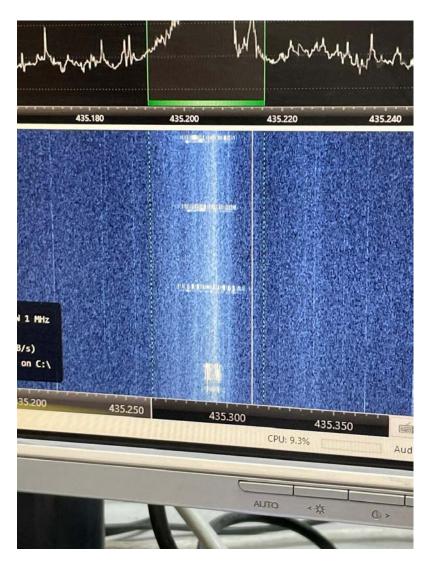


□ <u>SPINO</u>









Conclusions

Our main scientific goal is:

- To observe essential climate variables with a constellation of small satellites.

The INSPIRE goals are:

- To initiate a Space Program, and to teach courses related to Space.
- To have Laboratory facilities for hardware development and specialized personnel for teaching.
- To have facilities for building and testing CubeSat/small Instruments.
- To have ground stations for satellite operations.

Our positions are:

- To Design for simplicity and robustness:
 - Assume designs will fail and then prove they will work.
 - Design the satellite for easy assembly and disassembly.
 - Have respectable margins, robust safe modes, few deployables, graceful
 - performance. degradation, and frequent preventative satellite resets.
- To Build an experienced team—it matters:

 A successful team has veteran member(s) and frequent informal peer reviews (discussions) with proven subject matter experts.

Importance to implement one of our ground station (ELSA or HERMES) in La Réunion.

Conclusions

